

Lecture 9-10

- **- Flexural Members**
- ✓ **-**I- Laterally Restrained Beams
- ✓ II- Laterally Unrestrained Beams
- **Beam-Column Members**

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- **Axial force members are, in practice, subjected to axial load as well as bending in either or both the axis of the cross section.**
- **Similarly flexural members may also be subjected to axial load.**
- **In either case, a member subjected to both significant axial and bending stresses is termed as Beam-Column Members.**
- **The behavior of such members results from the combination of both effects and varies with slenderness.**

A member subjected to both significant axial and bending stresses is termed as Beam-Column Members.

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- **At low slenderness, the cross sectional resistance dominates.**
- **With increasing slenderness, pronounced secondorder effects appear, significantly influenced by both geometrical imperfections and residual stresses.**
- **At high slenderness range, buckling is dominated by elastic behavior, failure tending to occur by flexural buckling (typical of members in pure compression) or by lateral-torsional buckling (typical of members in bending).**
	- **The behavior of a member under bending and axial force results from the interaction between instability and plasticity and is influenced by geometrical and material imperfections. Therefore very complex.**

The verification of the safety of members subject to bending and axial force is made in two steps:

- **Verification of the resistance of cross sections .**
- **Verification of the member buckling resistance (in general governed by flexural or lateral-torsional buckling).**

Cross section resistance

The cross section resistance is based;

- **on its plastic capacity (class 1 or 2 sections) or**
- **on its elastic capacity (class 3 or 4 cross sections).**

When a cross section is subjected to bending moment and axial force $(N + M_y, N + M_z)$ or even $N + M_y + M_z$),

the bending moment resistance should be reduced, using interaction formulas.

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- **The interaction formulae to evaluate the elastic cross section capacity are the well known formulae of simple beam theory, valid for any type of cross section.**
- **However, the formulae to evaluate the plastic cross section capacity are specific for each cross section shape.**
- **For a cross section subjected to N + M, a general procedure may be established to evaluate the plastic bending moment resistance MN.Rd, reduced by the presence of an axial force N.**

• **Although the interaction formulae are easy to obtain by applying the general method, the resulting formulae differ for each cross sectional shape and are often not straightforward to manipulate.**

• **Historically, several approximate formulae have been developed, and, Villette (2004) proposed an accurate general formula, applicable to most standard cross sections. with an axis of symmetry with respect to the axis of bending, given by:**

$$
\frac{M_{Ed}}{M_{pl, Rd}} + \left(\frac{N_{Ed}}{N_{pl, Rd}}\right)^{\alpha, plan} = 1.0 \quad \alpha_{plan} = 1.0 + 1.82 \sqrt{\left(\frac{k}{w_{pl}} - 1.01\right) \frac{k-1}{w_{pl}}}
$$

- **wpl =Wpl/W^e is the ratio between the plastic bending modulus and the elastic modulus,** • **k=v/i is the ratio between the maximum distance v from an extreme fiber to the elastic**
- **neutral axis and the radius of gyration i of the section about the axis of bending.**

-
- **For a circular hollow section, the following exact expression may be established (Lescouarc'h, 1977): :**

$$
M_{N, Rd} = M_{pl, Rd} \sin \frac{\pi (1 - n)}{2} \text{ where, } n = N_{Ed} / N_{pl, Rd}
$$

• **Interaction formulae for axial force and bi-axial bending have usually the following general format:**

$$
\left[\frac{M_{y,Ed}}{M_{N,y,Rd}}\right]^\alpha + \left[\frac{M_{z,Ed}}{M_{N,z,Rd}}\right]^\beta = 1
$$

For I or H cross sections subjected to N + My + Mz,

$$
\alpha = (1.0 - 0.5\sqrt{n})\alpha_{y,plan}
$$

$$
\beta = \frac{1+n}{1.0-n^{\left(\alpha_{z,plan}-0.5\right)}},
$$

For RHS cross subjected to N + My + Mz,

$$
\alpha = \beta = \frac{1.7}{1 - 1.13 n^2} \quad \text{(if } n < 0.8 \text{)};
$$

$$
\alpha = \beta = 6 \quad \text{(if } n \ge 0.8 \text{)}.
$$

EC1993-1-1 Provisions

Clause 6.2.9 provides several interaction formulae between bending moment and axial force, in the plastic range and in the elastic range. These are applicable to most cross sections. But in all case the following shall be satisfied;

Class 1 or 2 sections

$$
M_{Ed} \leq M_{N,Rd}
$$

 M_{Ed} is the design bending moment and $M_{\text{N.Rd}}$ represents the design plastic moment resistance reduced due to the axial force N_{Ed}

For rectangular solid sections under uni-axial bending and axial force, $M_{N,Rdis}$ given by

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For low values of axial force, the reduction of the plastic moment resistance is not significant, as can be seen.

For doubly symmetric I or H sections,

 \blacktriangleright It is not necessary to reduce the plastic moment resistance about y if the two following conditions are satisfied:

$$
N_{Ed} \le 0.25 N_{pl, Rd}
$$
 and $N_{Ed} \le 0.5 h_w t_w f_y / \gamma_{M0}$

It is not necessary to reduce the plastic moment resistance about z if the following condition is verified:

 $M_{N, Rd} = M_{pl, Rd} (1 - n^{1.7})$

$$
N_{Ed} \le h_w t_w f_y / \gamma_{M0}
$$

For I or H sections, rolled or welded, with equal flanges and where fastener holes are not to be accounted for,

$$
M_{N,y,kd} = M_{pl,y,kd} \frac{1-n}{1-0.5 a} \quad \text{but } M_{N,y,kd} \le M_{pl,y,kd};
$$

$$
M_{N,z,Rd} = M_{pl,z,Rd}
$$
 if $n \le a$;
\n
$$
M_{N,z,Rd} = M_{pl,z,Rd} \left[1 - \left(\frac{n-a}{1-a} \right)^2 \right]
$$
 if $n > a$,
\nwhere, $a = (A-2bt_f)/A$, but $a \le 0.5$.

For circular hollow sections,

For RHS of uniform thickness and for welded box sections with equal flanges and equal webs and where fastener holes are not to be accounted for.

 $M_{N,y,Rd} = M_{pl,y,Rd} \frac{1-n}{1-0.5 a_w}$ but $M_{N,y,Rd} \le M_{pl,y,Rd}$ where $a_w \le 0.5$ and $a_f \le 0.5$ are the ratios between the area of the webs and of the flanges, respectively, and the gross area of the webs and of the flanges, respe

In a cross section under bi-axial bending and axial force, the $N + M_y + M_z$ interaction can be checked by the following condition:

$$
\left[\frac{M_{y,Ed}}{M_{N,y,Rd}}\right]^{\alpha} + \left[\frac{M_{z,Ed}}{M_{N,z,Rd}}\right]^{\beta} \le 1.0 \quad \text{and } \beta \text{ are parameters that depend on the shape of the cross section}
$$
\n
$$
\alpha = 2; \ \beta = 5n, \text{ but } \beta \ge 1;
$$
\ncircular hollow sections\n
$$
\alpha = \beta = 2;
$$
\nrectangular hollow sections\n
$$
\alpha = \beta = \frac{1.66}{1 - 1.13 n^2}, \text{ but } \alpha = \beta \le 6.
$$

Class 3 or 4 cross sections

In class 3 or 4 cross sections, the interaction between bending and axial force requires that the following condition be checked:. f_{ν}

$$
\sigma_{x,Ed} \leq \frac{\sigma}{\gamma_{M0}}
$$

where

 $\sigma_{x,Ed}$ is the design value of the local longitudinal stress due to bending moment and axial force, taking into account the fastener holes where relevant. It is calculated based on the gross cross section for class 3 cross sections, and on a reduced effective cross section for class 4 sections.

Interaction of bending, axial and shear force

The interaction between bending, axial and shear force should be checked as follows:

- When $V_{Ed} \leq$ % 50 of the design plastic shear resistance $V_{PI, Rd}$, no reduction need be made in the bendin and axial force resistances
- When V_{Ed} > % 50 of the design plastic shear resistance $V_{PI, Rd}$, then the design resistance to the combination of bending moment and axial force should be calculated using a reduced yield strength for the shea area. This reduced strength is given by $(1-p)f_y$, where $p=(2V_{Ed}/V_{Pl, Rd}-1)^2$

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Buckling Resistance: Introduction

For a member under bending and compression, besides the first-order moments and displacements (obtained based or the undeformed configuration), additional second-order moments and displacements exist ("P-8" effects); these should be taken into account.

In the past, various interaction formulae have been propo to represent this situation over the full slenderness range.

The present approach of EC3-1-1 is based on a linear-addi interaction formula, illustrated by expression:

$$
f(\frac{N}{N_u}, \frac{M_y}{M_{uy}}, \frac{M_z}{M_{uz}}) \le 1.0
$$

Where,

N, M_v and M_z are the applied forces and N_u , M_{uv} and M_{uz} are the design resistances, that take in due acco the associated instability phenomena.

The development of the design rules, and in particular those adopted by EC3-1-1, is quite complex, as they have to incorporate;

- \triangleright two instability modes, flexural buckling and lateral-torsional buckling (or a combination of both),
- In different cross sectional shapes and several shapes of bending moment diagram, among other aspects.
- several common concepts, such as that of equivalent moment, the definition of buckling length and the concept of amplification.

Several procedures provided in EC3-1-1 were described for the verification of the global stability of a steel structure, including the different ways of considering the second order effects (local P- δ effects and global P- Δ effects).

This topic is solely focused on dealing with the second order effect arising from local $P-\delta$ effects.

Local P- δ effects are generally taken into account according to the procedures given in clause 6.3 of EC3-1-1

Clause 6.3.3(1) considers two distinct situations

such as members of circular hollow section or other sections restrained from torsion.

Here, flexural buckling is the relevant instability mode.

Members that are susceptible to torsional deformations,

such as members of open section (I or H sections) that are not restrained from torsion.

Here, lateral torsional buckling tends to be the relevant instability mode.

In EC3-1-1 two methods are given for the calculation of the interaction factors k_{w} , k_{w} , k_{z} and k_{zz} .

Regardless of the method to be applied;

In members that are not susceptible to torsional deformation, it is assumed that there is no risk of lateral torsional buckling (x_{1T} = 1.0). And calculating the interaction factors k_{yy} , k_{yz} , k_{zy} and k_{zz} for a member not susceptible to torsional deformation.

Method 1, developed by a group of French and Belgian researchers,

According to this method, a member is not susceptible to torsional deformations if
\n
$$
\begin{aligned}\n&\downarrow_{\Gamma} \geq I_{\gamma} \text{ or} \\
&\downarrow_{\Gamma} \leq I_{\gamma} \text{ but the following condition is satisfied.} \\
&\bar{\lambda}_0 \leq 0.2 \sqrt{C_1} \sqrt[4]{\left(1 - \frac{N_{Ed}}{N_{cr,z}}\right)} \left(1 - \frac{N_{Ed}}{N_{cr,z}}\right),\n\end{aligned}
$$

Where,

C₁ is a coefficient that depends on the shape of the bending moment diagram between laterally braced sections N_{cr} and N_{cr} represent the elastic critical loads for flexural buckling about z and for torsional buckling, respectively λ_0 is the non-dimensional slenderness coefficient for lateral torsional buckling, assessed for a situation with constant bending moment.

Method 1, developed by a group of French and Belgian researchers, Annex A of EC3-1-1 presents Tables, for the calculation of the interaction factors according to Method 1

$$
C_{yy} = 1 + (w_y - 1) \left[\left(2 - \frac{1.6}{w_y} C_{my}^2 \overline{\lambda}_{max} - \frac{1.6}{w_y} C_{my}^2 \overline{\lambda}_{max}^2 \right) n_{pl} - b_{LT} \right] \ge \frac{W_{el,y}}{W_{pl,y}}
$$

\nwhere $b_{LT} = 0.5 a_{LT} \overline{\lambda}_{e}^2 \frac{M_{y,Ed}}{\chi_{LT} M_{pl,y,kd}} \frac{M_{z,Ed}}{M_{pl,z,kd}}.$
\n
$$
C_{yz} = 1 + (w_z - 1) \left[\left(2 - 14 \frac{C_{mx}^2 \overline{\lambda}_{max}^2}{w_z^5} \right) n_{pl} - c_{LT} \right] \ge 0.6 \sqrt{\frac{w_z}{w_y}} \frac{W_{el,z}}{W_{pl,z}},
$$

\nwhere $c_{LT} = 10 a_{LT} \frac{\overline{\lambda}_{e}^2}{5 + \overline{\lambda}_{e}^4} \frac{M_{y,Ed}}{C_{my} \chi_{LT} M_{pl,y,Rd}}.$
\n
$$
C_{zy} = 1 + (w_y - 1) \left[\left(2 - 14 \frac{C_{my}^2 \overline{\lambda}_{max}^2}{w_y^5} \right) n_{pl} - d_{LT} \right] \ge 0.6 \sqrt{\frac{w_y}{w_z}} \frac{W_{el,y}}{W_{pl,y}},
$$

\nwhere $d_{LT} = 2 a_{LT} \frac{\overline{\lambda}_{0}}{0.1 + \overline{\lambda}_{e}^4} \frac{M_{y,Ed}}{C_{my} \chi_{LT} M_{pl,y,Rd}} \frac{M_{z,Ed}}{C_{mx} M_{pl,z,Ra}}.$
\n
$$
C_{zz} = 1 + (w_z - 1) \left[\left(2 - \frac{1.6}{w_z} C_{mx}^2 \overline{\lambda}_{max} - \frac{1.6}{w_z} C_{mx}^2 \overline{\lambda}_{max}^2 \right) - e_{LT} \right] n_{pl} \ge \frac{W_{el,z}}{W_{pl,z}},
$$

\nwhere $e_{LT} = 1.7 a_{LT} \frac{\overline{\lambda}_{0}}{0.1 + \overline{\lambda}_{e}^4} \frac{M_{y,Ed}}{C_{my} \chi_{LT} M_{pl,y,Rd}}.$

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Annex A of EC3-1-1 presents Tables, for the calculation of the interaction factors according to Method 1

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Annex A of EC3-1-1 presents Tables, for the calculation of the interaction factors according to Method 1 Auxiliary terms (continuation):

 $\overline{\lambda}_{\max} = \max(\overline{\lambda}_{\nu}, \overline{\lambda}_{z});$ $\varepsilon_y = \frac{M_{y, Ed}}{N_{Ed}} \frac{A}{W_{ely}}$ for class 1, 2 or 3 cross sections; $\overline{\lambda}_0$ = non dimensional slenderness for lateral torsional buckling due to uniform ending moment, that is, taking Ψ _v =1.0 in Table 3.15; $\mathcal{E}_y = \frac{M_{y, Ed}}{N_{\pi\omega}} \frac{A_{\text{eff}}}{W_{\text{eff}}}$ for class 4 cross sections; λ_{LT} = non dimensional slenderness for lateral torsional buckling; If $\overline{\lambda}_0 \le 0.2\sqrt{C_1} \sqrt[4]{\frac{1-N_{Ed}}{N_{cr,z}}\sqrt{1-\frac{N_{Ed}}{N_{cr,z}}}}$: $C_{my} = C_{my,0}$; $C_{mz} = C_{mz,0}$; $C_{mL} = 1.0$; $\frac{N_{cr,y}}{N_{cr,z}}$ is the elastic critical load for flexural buckling about y; $N_{cr,z}$ is the elastic critical load for flexural buckling about z; $N_{cr,T}$ is the critical load for torsional buckling;
 I_T is the constant of uniform torsion or St. Venant's torsion;
 I_y is the second moment of area about y; If $\overline{\lambda}_0 > 0.2 \sqrt{C_1} \sqrt[4]{\left(1 - \frac{N_{Ed}}{N_{max}}\right)\left(1 - \frac{N_{Ed}}{N_{max}}\right)}$. $C_{my} = C_{my,0} + \left(1 - C_{my,0}\right) \frac{\sqrt{\varepsilon_y} a_{LT}}{1 + \sqrt{\varepsilon_y} a_{LT}}$. $C_{mz} = C_{mz,0}$; $C_{mLT} = C_{my}^2 \frac{a_{LT}}{\sqrt{\left(1 - \frac{N_{Ed}}{N_{cm}}\right)\left(1 - \frac{N_{Ed}}{N_{cm}}\right)}} \ge 1$; $C_1 = \left(\frac{1}{k_c}\right)^2$ where k_c is taken from Table 3.10.

Method 2, developed by a group of Austrian and German researchers,

According to Method 2, the following members may be considered as not susceptible to torsional deformation:

 \triangleright members with circular hollow sections (CHS).

- members with rectangular hollow sections (RHS) (there is widlly argued exception to this rule presented in (I
- members with open cross section, provided that they are torsionally and laterally restrained.
- Annex B of EC3-1-1 presents Tables, for the calculation of the interaction factors according to Method 2

Method 2, developed by a group of Austrian and German researchers,

Annex B of EC3-1-1 presents Tables, for the calculation of the interaction factors according to Method 2

Design According to EC3: Section classification for sections under bending and axial force

According to EC3, the classification of a cross section is based on its maximum resistance to the type of applied internal forces, independent from their values.

- This procedure is straightforward to apply for cross sections subjected to either bending or compression.
- However, the presence of both the compression and bending moment on the cross-section member, generates a stress distribution between that related to pure compression and that associated with the presence of the sole bending moment.
- Expering in mind this additional complexity, simplified procedures are often adopted, such as:
- to consider the cross section subjected to compression only, being the most unfavourable situation (too conservative in some cases)
- to classify the cross section based on an estimate of the position of the neutral axis based on the applied ii. internal forces.
- \triangleright In the later case the neutral axis depth depends on whether the section can plastify, the bending axis, the section profile.

Design According to EC3: Section classification for sections under bending and axial force

With reference to the case of a neutral axis located in the web, α ranges between 0.5 (bending) and 1 (compression) and ψ ranges between -1 (bending) and 1 (compression).

Once the stress distribution is assumed and the values of α and ψ can be used to classify the section using tables 5.2 (sheet1 through 3)